

# **MEOSAR Overview**

SAR Controllers Training 2014 24 – 28 March 2014

> Chris O'Connors NOAA







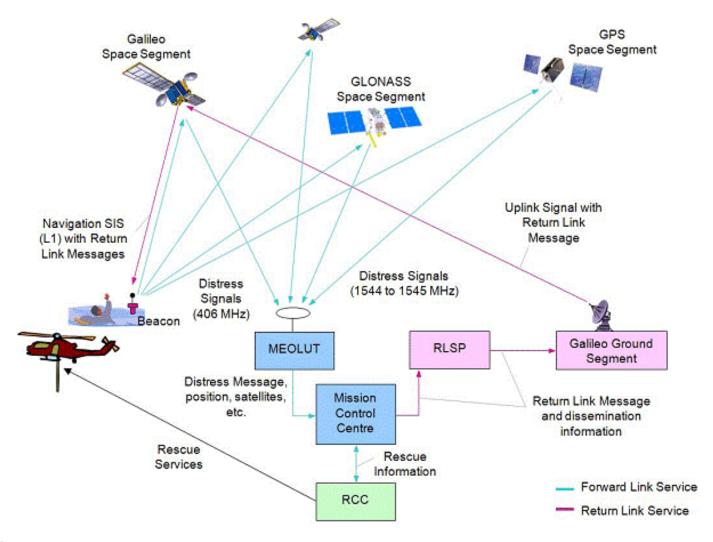
# Agenda



- MEOSAR/DASS Overview
- Space Segment
- Ground Segment
- DASS POC
- MEOSAR Timeline
- Demonstration and Evaluation



# MEOSAR/DASS Overview

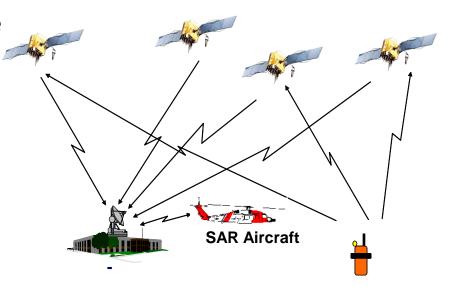


# MEOSAR/DASS Overview



## MEOSAR/DASS Concept

- Utilize multiple satellites with SAR repeaters, or "bent pipe"
- Multiple antennas are used to receive the same beacon burst
- The time difference of arrival (TDOA) and frequency difference of arrival (FDOA) is then used to determine location
- One burst, received through 3 unique satellites, is capable of producing a location
- Essentially, GPS location in reverse



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# MEOSAR/DASS Overview



# Medium Earth Orbit (MEO) SAR / Distress Alerting Satellite System (DASS)

- Various studies determined that medium-earth orbiting (MEO) satellites provide a vastly improved space-based distress alerting and locating system.
- NASA, with USAF Space and Air Combat Command, NOAA, and USCG are developing a capability on GPS satellites— Distress Alerting Satellite System (DASS)

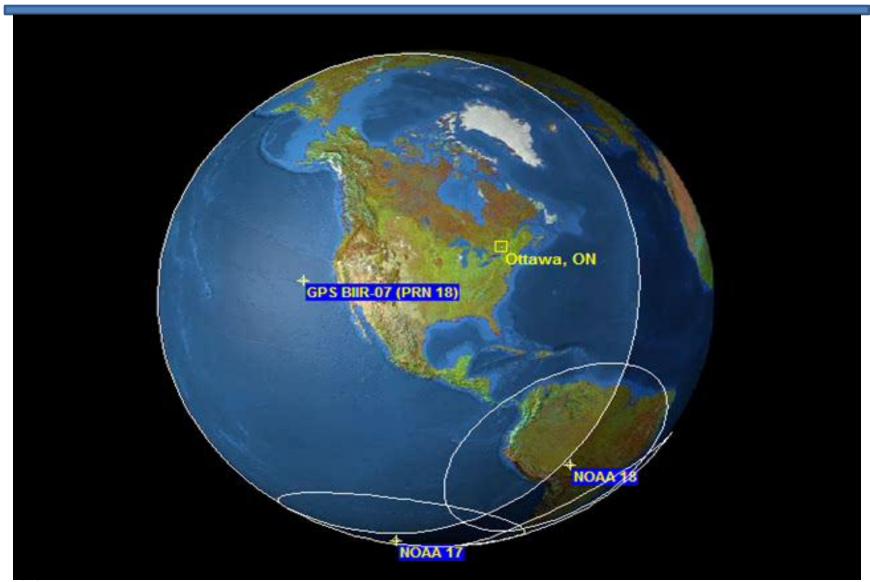
## MEOSAR provides

- A combination of the best assets of GEOSAR and LEOSAR
- Near instantaneous notification and location of distress
- Near 100% Availability
- Better location accuracy
- Global coverage
- Full compatibility with current and future beacons

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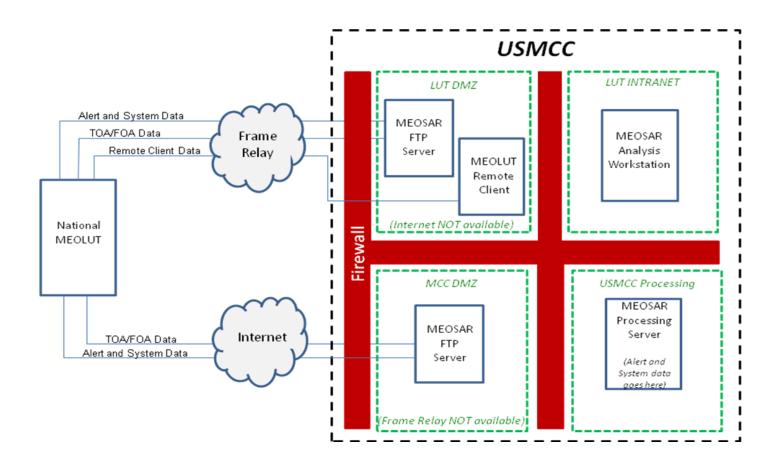


# MEO vs. LEO Coverage





# **US MEOSAR Ground Segment Design**





# **Ground Segment**

- Prototype ground station at NASA Goddard Space Flight Center
  - 4 antennas capable of independently tracking 4 satellites
  - Completed in 2008
  - Successfully passed acceptance testing
  - May become future operational MEOLUT
  - Full participation in MEOSARD&E testing



# **Ground Segment**



- Accepted MEOLUT Wahiawa, Hawaii
  - 6 antenna capable of tracking 6 satellites either S-band or L-band
  - Constructed in September 2011 and passed acceptance testing



# **Ground Segment**



- Proposed MEOLUT in Miami, Florida
  - 6 antenna capable of tracking 6 satellites either S-band or L-band
  - CDR completed November 2013
  - Construction and site acceptance tests will be completed by March 1<sup>st</sup> 2014



# DASS Proof-of-Concept

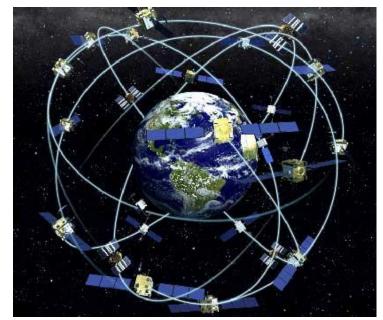


- DASS Proof-of-Concept (POC) Space Segment
  - Thirteen current on-orbit GPS Block IIR and IIF satellites carry DASS repeaters
  - POC system uses existing GPS. Downlink at S-Band (Not ITU-allocated for SAR, but may possibly be used operationally)
- Proof-of-Concept results to date:
  - Demonstrated ability to locate beacons to greater than current Cospas-Sarsat accuracy using two or more satellites
  - System meets/exceeds theoretical capabilities
  - Tests are on-going

# Space Segment



- Repeaters will be flown on Medium Earth Orbit satellites (MEOSATs)
- Will utilize 3 Global Navigation Satellite System (GNSS) constellations
  - GPS (USA)
  - GLONASS (Russia)
  - Galileo (ESA)
- Current plan is to have 24
  US MEOSAR instruments
- 72 MEOSAR instruments total





# Search and Rescue on GPS (SAR/GPS)



- NASA, with USAF Space and Air Combat Command, NOAA, and USCG are developing a capability on GPS satellites— SAR/GPS
- SAR/GPS provides:
  - 406 MHz "bent-pipe" repeaters on GPS Alert data downlink freely available internationally in L-band
  - Full compatibility with existing and future 406 MHz beacons
  - Support for civilian and military SAR responsibilities
  - Low technical risk, low cost
  - SAR/GPS fully interoperable with similar proposed Russian (SAR/GLONASS) and European (SAR/Galileo) systems
- Supports U.S. Policy on Space-Based Positioning, Navigation & Timing Policy – S&R on GPS

# **MEOSAR Transition**



2011	2012	2013	2014	2015	2016	2017	2018	2019	2020
Second Generation Beacons									
MEOSAR Ground System Development									
1	MESOAR	D&E	F	re-IO	c	IOC		FO	C
						Galile	O		
	GLONASS								
									SAR/GPS
				DASS					SAR/GP3

### **US MEOSAR Timeline**



- 2011 planning for MEOSAR D&E testing;
- 2012 participant ground segment preparation for D&E testing;
- 2013 Phase I MEOSAR D&E technical testing;
- 2014 Phase II MEOSAR D&E operational and technical testing;
- 2015 Phase III MEOSAR D&E L-band operational and technical testing; and
- 2015 compilation of test results and analysis, and preparation of final D&E report.

# MEOSAR Demonstration and Evaluation (D&E)



### Goals

- Characterize technical and operational performance
- Evaluate operational effectiveness
- Provide basis for recommendations on the integration of MEOSAR system into C/S

Basis for commissioning criteria

# MEOSAR Demonstration and Evaluation (D&E)



### Technical tests

- Processing threshold and system margin
- Impact of interference
- Valid and complete message acquisition
- Location accuracy
- System Capacity
- Networked MEOLUT advantage
- Combined MEO/GEO performance

# MEOSAR Demonstration and Evaluation (D&E)



- Operational Tests
  - Time advantage
  - Unique detections
  - Volume of ground segment traffic
  - SAR/Galileo RLS
  - Direct and indirect benefits of MEOSAR system
  - Validate Data distribution procedures
  - Validate quality factor(s) for location

# C/S R.018 – Section C.1.3.14 (2 of 2)



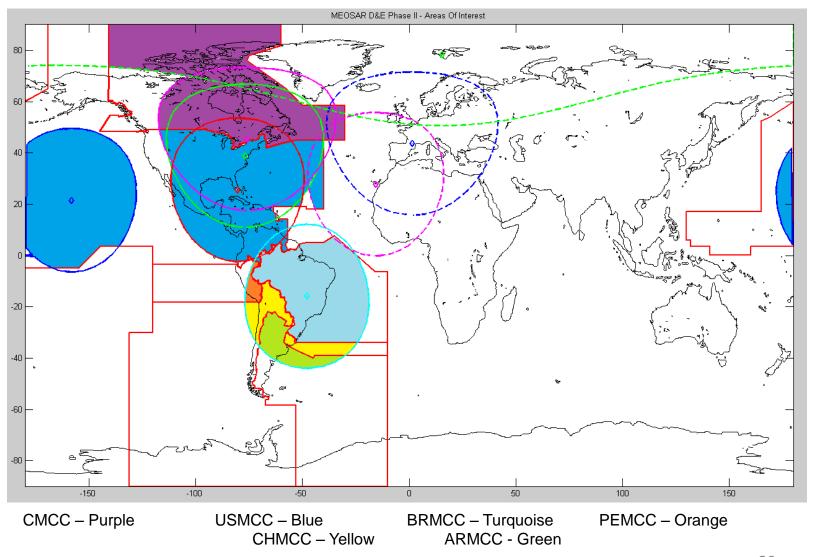
Performance Parameter	Requirement	Definition	Reference in LEOLUT Performance Specification C/S T.002
Detection Probability	99%	The probability of detecting the transmission of a 406 MHz beacon and recovering at the MEOLUT a valid beacon message, within 10 minutes from the first beacon message transmission.	Paragraph 5.3.2
Independent Location Probability	98%	The probability of obtaining at the MEOLUT a 2D location (Lat/Long), independently of any encoded position data in the 406 MHz beacon message, within 10 minutes from the first beacon message transmission.	Paragraph 5.3.3
Independent Location Error	P(e < 5 km) > 95%	The system independent location solution should be within 5 km from the actual beacon position 95% of the time.	Paragraph 5.3.6
Single Burst Independent Location Probability	90%	The probability of obtaining at the MEOLUT a 2D location (Lat/Long), independently of any encoded position data in the 406 MHz beacon message, using a single burst transmission.	
Single Burst Independent Location Error	P(e < 5 km) > 90%	The Single Burst independent location solution should be within 5 km from the actual beacon position 90% of the time.	
Sensitivity	≤34.8 dB-Hz	The minimum C/No level at which the MEOLUT will correctly process beacon messages, where C/No is the ratio of the unmodulated carrier power to noise power density in dB-Hz. The value is that necessary to obtain a 5x10 <sup>-5</sup> BER for the beacon message.	Paragraph 5.3.1

Table C.1: MEOLUT Minimum Performance Requirements

## Introduction

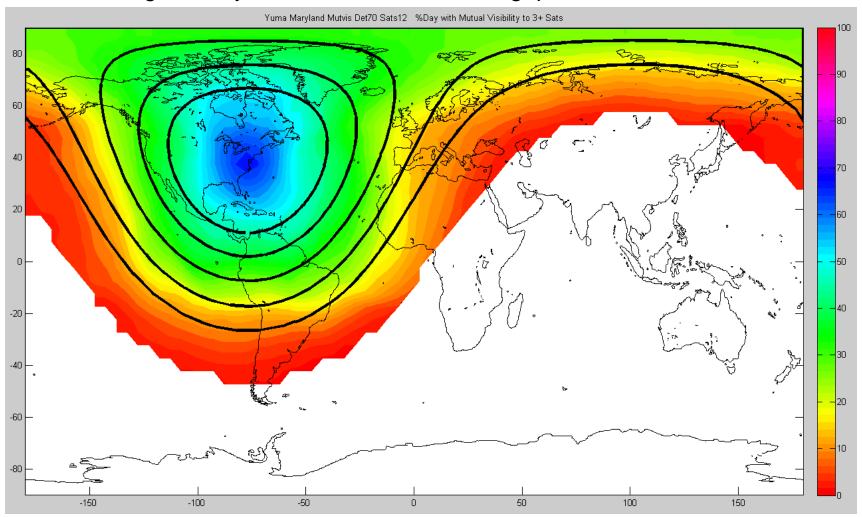


#### AOI for Location – All WDDR MCCs



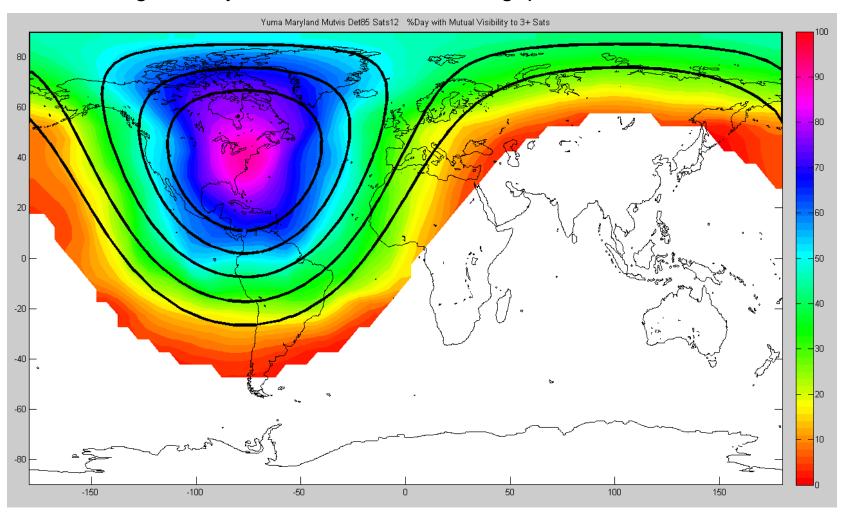


#### Percentage of Day with 3+ Satellites: Throughput 70% and 12 Satellites



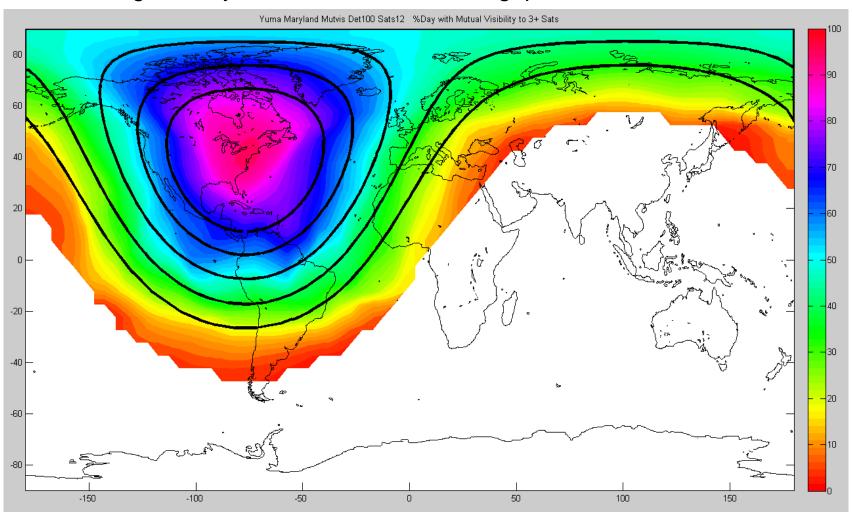


#### Percentage of Day with 3+ Satellites: Throughput 85% and 12 Satellites



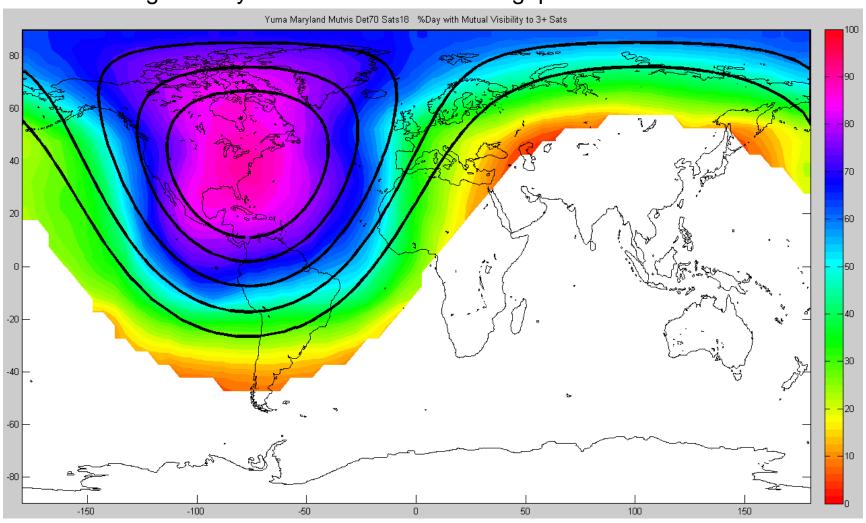


#### Percentage of Day with 3+ Satellites: Throughput 100% and 12 Satellites



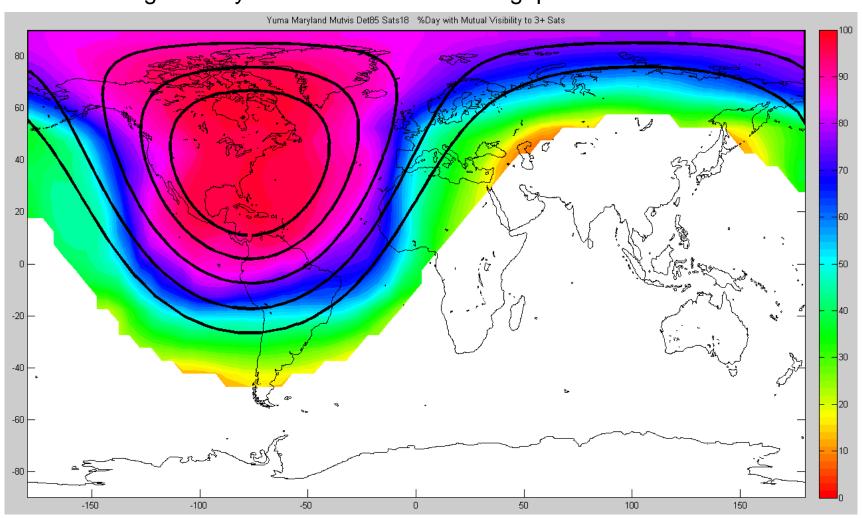


#### Percentage of Day with 3+ Satellites: Throughput 70% and 18 Satellites



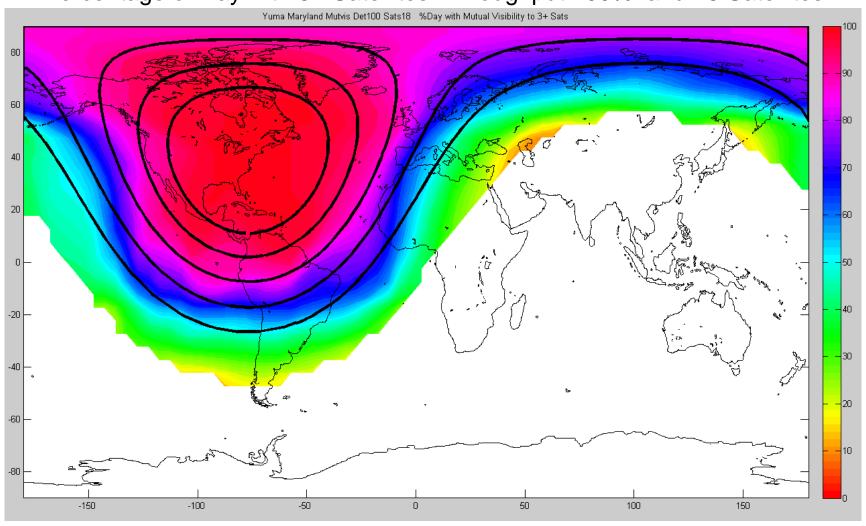


#### Percentage of Day with 3+ Satellites: Throughput 85% and 18 Satellites



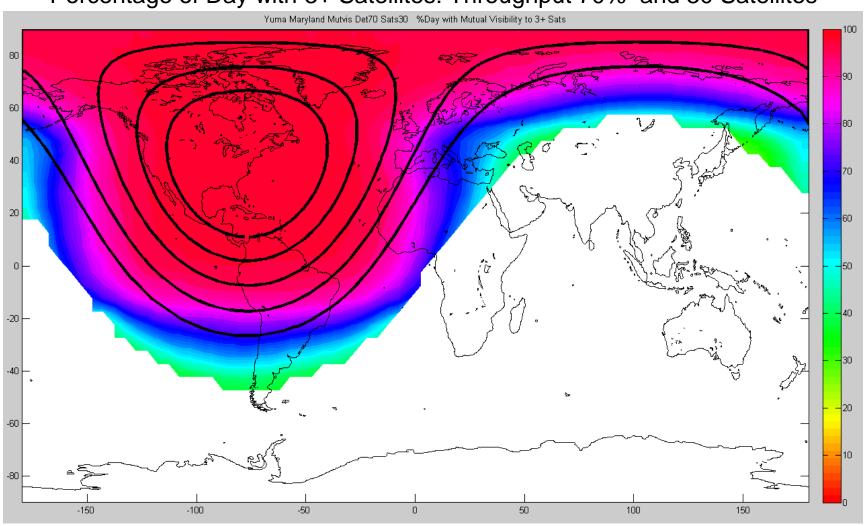


#### Percentage of Day with 3+ Satellites: Throughput 100% and 18 Satellites





#### Percentage of Day with 3+ Satellites: Throughput 70% and 30 Satellites





#### Percentage of Day with 3+ Satellites: Throughput 85% and 30 Satellites

